$$K_{f} = \begin{bmatrix} 4.3271e + 00 & 1.2987e + 00 & 6.7886e - 01 & 5.1942e - 01 \\ 2.5915e + 01 & 3.0218e + 00 & 6.7930e - 01 & 8.3097e - 01 \\ 1.3071e + 01 & 1.8009e + 00 & 9.3469e - 01 & 3.4572e - 02 \\ 1.4255e + 01 & 1.5307e + 00 & 3.8350e - 01 & 5.3462e - 02 \end{bmatrix}$$

$$(45)$$

$$N = \begin{bmatrix} 2.0750e + 05 & -1.4364e + 05 & -2.2264e + 05 & 4.2166e + 05 \\ -1.3105e + 05 & 9.0701e + 04 & 1.4062e + 05 & -2.6628e + 05 \end{bmatrix}$$
(46)

$$M = \begin{bmatrix} 4.1057e + 02 & -5.2786e + 00 & 6.3758e + 02 & -1.8704e + 02 \\ -2.5469e + 02 & -1.0739e + 02 & -3.2603e + 02 & 1.1764e + 02 \end{bmatrix}$$
(47)

The minimum return difference matrix of control system, loop transfer function inf  $\sigma[I+C(s)P(s)]=\beta=0.59$ . Therefore, the multivariable stability margins, <sup>24</sup> gain margin and phase margin, are precisely the same as those of the regulator when the observer is included in the flight control system. Figure 3 shows the target loop and control system loop transfer function.

#### **Conclusions**

We have given an algorithm that achieves precise loop transfer recovery and provides freedom of eigenstructure assignment for nonsquare plants in a situation in which the number of sensors exceeds the number of controls. It is important to note that the algorithm yields finite observer gain, a critical requirement for pragmatic design. This approach is computationally simple, requiring on the order of  $n^3$  operations, and works for regular systems with no transmission zeros.

The flexibility in selection of observer eigenvalues can be used to meet other performance requirements. In particular, in the case of flight control problems this flexibility can be used to meet handling and flying quality requirements. Much work can be done in the area of exploring the selection of observer poles needed to achieve certain desired handling qualities and better performance in general.

The situation in which the number of actuators exceeds the number of sensors is dual to this case, and corresponds to loop transfer recovery at the output. This dual version of loop transfer recovery has been explained by a number of researchers.<sup>4,8,11</sup> Tsui<sup>8</sup> states that the necessary and sufficient conditions for dual loop transfer recovery are CT = 0 and  $AT - TF = BK_c$ . In this case, the state feedback system does not have a reduced-order version. There exists no full rank matrix T to exactly satisfy these dual conditions.<sup>14</sup> One can conclude that, in this case, precise loop transfer recovery is not possible.

### **Appendix**

#### Proof of Theorem 1

The assumption on the rank of CB is sufficient to guarantee the existence of the QR factors in steps 1 and 3 and the invertibility of  $R_1$ . The distinct eigenvalue hypothesis guarantees that the Sylvester equation in step 5 has a unique solution. Thus, all of the indicated computations in steps 1-6 can be performed. We now verify that we have satisfied Eqs. (11) and (10). In step 6, we set  $T = ZW_2^T$ , so that  $TB = (ZW_2^T)(W_1S_1) = 0$ , since  $W_2^TW_1 = 0$ , and so Eq. (11) is satisfied. Now, the matrix T satisfies the Sylvester equation (10) if and only if

$$ZW_2^T A - FZW_2^T = K_f C (A1)$$

or, when we multiply by the nonsingular matrix W,

$$ZW_2^T AW - FZW_2^T W = K_f CW (A2)$$

Equation (A2) can be rewritten as

$$Z[A_1 \quad A_2] - FZ[0 \quad I] = K_f[C_1 \quad CW_2]$$
 (A3)

which is equivalent to the two conditions

$$ZA_1 = K_f C_1 \tag{A4}$$

and

$$ZA_2 - FZ = K_f CW_2 \tag{A5}$$

We now verify that the matrices Z and  $K_f$  determined by the algorithm satisfy these relations. By step 6,

$$K_f C_1 = [ZA_1R_1^{-1} \ L_2]Q^T C_1 = [ZA_1R_1^{-1} \ L_2] \begin{bmatrix} R_1 \\ 0 \end{bmatrix} = ZA_1$$
(A6)

and by steps 5, 4, and 6,

$$ZA_2 - FZ = ZA_1R_1^{-1}E_1 + L_2E_2$$
 (A7)

$$= [ZA_1R_1^{-1} \quad L_2]\begin{bmatrix} E_1 \\ E_2 \end{bmatrix} \tag{A8}$$

$$= [ZA_1R_1^{-1} L_2]Q^TCW_2$$
 (A9)

$$=K_f C W_2 \tag{A10}$$

as desired.

The second theorem concerns the meaning of transmission zeros. We will use several facts about the rank of a matrix. A matrix Y with at least as many rows as columns has full rank if and only if there exists no nonzero vector h such that Yh = 0. The rank of a matrix is unchanged if 1) the matrix is multiplied by a square, nonsingular matrix, 2) row operations are performed, adding a multiple of one row to another, and 3) rows or columns are reordered.

Finally, if

$$Y = \begin{bmatrix} Y_1 & Y_2 \\ 0 & Y_3 \end{bmatrix} \tag{A11}$$

where  $Y_1$  is square, then Y is full rank if and only if  $Y_1$  and  $Y_3$  are full rank.

### **Proof of Theorem 2**

We multiply the transmission matrix by square nonsingular matrices to produce the following product:

$$p \\ n-p \\ n \\ m \\ \begin{bmatrix} I & 0 & 0 \\ 0 & I & 0 \\ 0 & 0 & Q^T \end{bmatrix} p \\ n-p \\ m \\ \begin{bmatrix} W_1^T & 0 \\ W_2^T & 0 \\ 0 & I \end{bmatrix} n \\ m \\ \begin{bmatrix} A-sI & B \\ C & 0 \end{bmatrix} p \\ \begin{bmatrix} W_1 & W_2 & 0 \\ 0 & 0 & I \end{bmatrix}$$

After permutation, and subtraction of  $A_1R_1^{-1}$  times the third block of rows from the second block, we obtain the following matrix, whose rank is the same as that of the transmission matrix:

$$\begin{array}{c|ccccc}
p & & & & & & & & & & & & & & & & \\
p & & S_1 & W_1^T A W_1 - s I & & W_1^T A W_2 & & & & \\
p & & 0 & & R_1 & & E_1 & & & \\
n - p & & 0 & & 0 & & A_2 - A_1 R_1^{-1} E_1 - s I \\
m - p & & 0 & & & E_2
\end{array} \right) (A15)$$

Since, by regularity,  $R_1$  and  $S_1$  are full rank, we see that there are no transmission zeros if and only if the reduced system is observable.

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# References

<sup>1</sup>Doyle, J. C., "Guaranteed Margins for LQG Regulators," *IEEE Transactions on Automatic Control*, Vol. AC-23, Aug. 1978, pp. 756–757.

<sup>2</sup>Safonov, M. G., and Athans, M., "Gains and Phase Margin of Multiloop LQG Regulators," *IEEE Transactions on Automatic Control*, Vol. AC-22, April 1977, pp. 173-179.

<sup>3</sup>Doyle, J. C., and Stein, G., "Robustness with Observers," *IEEE Transactions on Automatic Control*, Vol. AC-24, Aug. 1979, pp. 607-611.

<sup>4</sup>Stein, G., and Athans, M., "The LQS/LTR Procedure for Multivariable Feedback Control Design," *IEEE Transactions on Automatic Control*, Vol. AC-32, Feb. 1987, pp. 105-114.

<sup>5</sup>Madiwale, A. N., and Williams, D. E., "Some Extensions of Loop Transfer Recovery," *Proceedings of the IEEE American Control Conference* (Boston, MA), Inst. of Electrical and Electronics Engineers, New York, 1985, pp. 790-795.

<sup>6</sup>Calise, A. J., and Prasad, J. V. R., "An Approximate Loop Transfer Recovery Method for Designing Fixed-Order Compensators," *Journal of Guidance, Control, and Dynamics*, Vol. 13, No. 2, 1990, pp. 297-302.

<sup>7</sup>Fu, M., "Exact, Optimal, and Partial Loop Transfer Recovery,"

<sup>7</sup>Fu, M., "Exact, Optimal, and Partial Loop Transfer Recovery," *Proceedings of the IEEE Decision and Control Conference* (Honolulu, HI), Inst. of Electrical and Electronics Engineers, New York, 1991, pp. 1841–1846.

<sup>8</sup>Tsui, C. C., "On the Loop Transfer Recovery," *Proceedings of the IEEE American Control Conference* (Pittsburgh, PA), Inst. of Electrical and Electronics Engineers, New York, 1989, pp. 2184–2189.

<sup>9</sup>Tsui, C. C., "New Approach to Robust Observer Design," *International Journal of Control*, Vol. 47, 1988, pp. 745-751.

<sup>10</sup>Luenberger, D. G., "Introduction to Observers," *IEEE Transactions on Automatic Control*, Vol. AC-16, Dec. 1971, pp. 596-602.

<sup>11</sup>Doyle, J. C., and Stein, G., "Multivariable Feedback Design: Concepts for Classical/Modern Synthesis," *IEEE Transactions on Automatic Control*, Vol. AC-26, Feb. 1981, pp. 4-16.

<sup>12</sup>Ridgely, B. D., and Banda, S. S., "Introduction to Robust Multivariable Control," Air Force Wright Aeronautical Lab., AFWALTR-85-3102, Wright-Patterson AFB, OH, Feb. 1986, pp. 2-34.

<sup>13</sup>Maciejowski, J. M., *Multivariable Feedback Design*, Addison-Wesley, Reading, MA, 1989, pp. 52-55.

14 Barlow, J. B., Monahemi, M. M., and O'Leary, D. P., "Constrained Matrix Sylvester Equations," SIAM Journal on Matrix Analysis and Applications, Vol. 13, Jan. 1992, pp. 1-9.

<sup>15</sup>Dongarra, J. J., Moler, C. B., Bunch, J. R., and Stewart, G. W., *LINPACK User's Guide*, Society for Industrial and Applied Mathematics, Philadelphia, PA, 1979.

<sup>16</sup>Bartels, R. H., and Stewart, G. W., "Algorithm 432, Solution of a Matrix Equation AX + XB = C," Communications of the ACM, Vol. 15, Sept. 1972, pp. 820-826.

<sup>17</sup>Chen, B. M., Saberi, A., and Ly, U-L., "Closed-Loop Transfer Recovery with Observer-Based Controllers—Part 1: Analysis," *Proceedings of the AIAA Guidance, Control, and Navigation Conference* (New Orleans, LA), AIAA, Washington, DC, 1991, pp. 1095-1109.

<sup>18</sup> Joseph, P. D., and Tou, J., "On Linear Control Theory," *Transactions of the AIEE*, Pt. II, Vol. 90, Sept. 1961, pp. 193-196.

<sup>19</sup>Friedland, B., Control System Design, McGraw-Hill, New York, 1986, pp. 298-301.

<sup>20</sup>Rynaski, E. G., "Flight Control System Design Using Robust Output Observers," AGARD-CP 321, 1982, pp. 14-1-14-8.

<sup>21</sup>Gilbert, M., "Dynamic Modeling and Active Control of Aeroelastic Aircraft," M.S. Thesis, Dept. of Aeronautics and Astronautics, Purdue Univ., West Lafayette, IN, May 1982.

<sup>22</sup>MATLAB User's Guide, MathWorks, Inc., South Natick, MA,

<sup>23</sup>Kautsky, J., Nichols, N. K., and Van Dooren, P., "Robust Pole Assignment in Linear State Feedback," *International Journal of Con-*

trol, Vol. 45, No. 5, 1985, pp. 1129-1155.

24Lehtomaki, N. A., "Practical Robustness Measures in Multivariable Control System Analysis," Ph.D. Dissertation, Dept. of Electrical Engineering and Computer Science, Massachusetts Inst. of Technology, Cambridge, MA, May 1981.